Interview Summary	Application No.	Applicant(s)
	10/736,035	LIU ET AL.
	Examiner	Art Unit
	Sara Addisu	3722
All participants (applicant, applicant's representative, PTO personnel):		
(1) <u>Sara Addisu</u> .	(3)	
(2) <u>Anna Lemke (612) 339-7461</u> .	(4)	
Date of Interview: <u>05 May 2006</u> .		
Type: a)⊠ Telephonic b)□ Video Conference c)□ Personal [copy given to: 1)□ applicant 2)□ applicant's representative]		
Exhibit shown or demonstration conducted: d)		
Claim(s) discussed: Ms. Lemke confirmed that the application is abandoned.		
Identification of prior art discussed:		
Agreement with respect to the claims f) was reached. g) was not reached. h) N/A.		
Substance of Interview including description of the general nature of what was agreed to if an agreement was reached, or any other comments:		
(A fuller description, if necessary, and a copy of the amendments which the examiner agreed would render the claims allowable, if available, must be attached. Also, where no copy of the amendments that would render the claims allowable is available, a summary thereof must be attached.)		
THE FORMAL WRITTEN REPLY TO THE LAST OFFICE ACTION MUST INCLUDE THE SUBSTANCE OF THE INTERVIEW. (See MPEP Section 713.04). If a reply to the last Office action has already been filed, APPLICANT IS GIVEN A NON-EXTENDABLE PERIOD OF THE LONGER OF ONE MONTH OR THIRTY DAYS FROM THIS INTERVIEW DATE, OR THE MAILING DATE OF THIS INTERVIEW SUMMARY FORM, WHICHEVER IS LATER, TO FILE A STATEMENT OF THE SUBSTANCE OF THE INTERVIEW. See Summary of Record of Interview		
requirements on reverse side or on attached sheet.		
	_).
Examiner Note: You must sign this form unless it is an	Sana	ADILS 5 15/06
Attachment to a signed Office action.		ature, if required

Summary of Record of Interview Requirements

Manual of Patent Examining Procedure (MPEP), Section 713.04, Substance of Interview Must be Made of Record

A complete written statement as to the substance of any face-to-face, video conference, or telephone interview with regard to an application must be made of record in the application whether or not an agreement with the examiner was reached at the interview.

Title 37 Code of Federal Regulations (CFR) § 1.133 Interviews Paragraph (b)

In every instance where reconsideration is requested in view of an interview with an examiner, a complete written statement of the reasons presented at the interview as warranting favorable action must be filed by the applicant. An interview does not remove the necessity for reply to Office action as specified in §§ 1.111, 1.135. (35 U.S.C. 132)

37 CFR §1.2 Business to be transacted in writing.

All business with the Patent or Trademark Office should be transacted in writing. The personal attendance of applicants or their attorneys or agents at the Patent and Trademark Office is unnecessary. The action of the Patent and Trademark Office will be based exclusively on the written record in the Office. No attention will be paid to any alleged oral promise, stipulation, or understanding in relation to which there is disagreement or doubt.

The action of the Patent and Trademark Office cannot be based exclusively on the written record in the Office if that record is itself incomplete through the failure to record the substance of interviews.

It is the responsibility of the applicant or the attorney or agent to make the substance of an interview of record in the application file, unless the examiner indicates he or she will do so. It is the examiner's responsibility to see that such a record is made and to correct material inaccuracies which bear directly on the question of patentability.

Examiners must complete an Interview Summary Form for each interview held where a matter of substance has been discussed during the interview by checking the appropriate boxes and filling in the blanks. Discussions regarding only procedural matters, directed solely to restriction requirements for which interview recordation is otherwise provided for in Section 812.01 of the Manual of Patent Examining Procedure, or pointing out typographical errors or unreadable script in Office actions or the like, are excluded from the interview recordation procedures below. Where the substance of an interview is completely recorded in an Examiners Amendment, no separate Interview Summary Record is required.

The Interview Summary Form shall be given an appropriate Paper No., placed in the right hand portion of the file, and listed on the "Contents" section of the file wrapper. In a personal interview, a duplicate of the Form is given to the applicant (or attorney or agent) at the conclusion of the interview. In the case of a telephone or video-conference interview, the copy is mailed to the applicant's correspondence address either with or prior to the next official communication. If additional correspondence from the examiner is not likely before an allowance or if other circumstances dictate, the Form should be mailed promptly after the interview rather than with the next official communication.

The Form provides for recordation of the following information:

- Application Number (Series Code and Serial Number)
- Name of applicant
- Name of examiner
- Date of interview
- Type of interview (telephonic, video-conference, or personal)
- Name of participant(s) (applicant, attorney or agent, examiner, other PTO personnel, etc.)
- An indication whether or not an exhibit was shown or a demonstration conducted
- An identification of the specific prior art discussed
- An indication whether an agreement was reached and if so, a description of the general nature of the agreement (may be by attachment of a copy of amendments or claims agreed as being allowable). Note: Agreement as to allowability is tentative and does not restrict further action by the examiner to the contrary.
- The signature of the examiner who conducted the interview (if Form is not an attachment to a signed Office action)

It is desirable that the examiner orally remind the applicant of his or her obligation to record the substance of the interview of each case. It should be noted, however, that the Interview Summary Form will not normally be considered a complete and proper recordation of the interview unless it includes, or is supplemented by the applicant or the examiner to include, all of the applicable items required below concerning the substance of the interview.

A complete and proper recordation of the substance of any interview should include at least the following applicable items:

- 1) A brief description of the nature of any exhibit shown or any demonstration conducted,
- 2) an identification of the claims discussed,
- 3) an identification of the specific prior art discussed,
- 4) an identification of the principal proposed amendments of a substantive nature discussed, unless these are already described on the Interview Summary Form completed by the Examiner,
- 5) a brief identification of the general thrust of the principal arguments presented to the examiner,

(The identification of arguments need not be lengthy or elaborate. A verbatim or highly detailed description of the arguments is not required. The identification of the arguments is sufficient if the general nature or thrust of the principal arguments made to the examiner can be understood in the context of the application file. Of course, the applicant may desire to emphasize and fully describe those arguments which he or she feels were or might be persuasive to the examiner.)

- 6) a general indication of any other pertinent matters discussed, and
- 7) if appropriate, the general results or outcome of the interview unless already described in the Interview Summary Form completed by the examiner.

Examiners are expected to carefully review the applicant's record of the substance of an interview. If the record is not complete and accurate, the examiner will give the applicant an extendable one month time period to correct the record.

Examiner to Check for Accuracy

If the claims are allowable for other reasons of record, the examiner should send a letter setting forth the examiner's version of the statement attributed to him or her. If the record is complete and accurate, the examiner should place the indication, "Interview Record OK" on the paper recording the substance of the interview along with the date and the examiner's initials.

subscripts run from 0 to 3 and Latin indexes from 1 to 3. The symbol \circ denotes the Euclidian inner product; while |a| is the length of the vector a in the Euclidian space. The symbol |a| represents the absolute value of a scalar a. In addition, \bullet stands for the Riemannian inner product, and $\|\xi\|$ is the Riemannian norm of the vector ξ corresponding to the tensor ξ^{α} or ξ_{α} . f is a set of functions. We specify t as a time coordinate variable, and τ , a proper time variable. $D/D\tau$ denotes a covariant derivative operator vs τ .

II. COMPARISONS OF THREE METHODS FOR COMPUTING LE IN GENERAL RELATIVITY

A. LE in classical physics

In classical physics the LE is to characterize the mean exponential rate of divergence of trajectories surrounding a given trajectory in the phase space. For a compact autonomous n-dimensional system

$$\dot{\boldsymbol{X}} = \boldsymbol{f}(\boldsymbol{X}) \tag{1}$$

with the solution $\boldsymbol{X}(t) = (\boldsymbol{x}, \dot{\boldsymbol{x}})$ and its corresponding variational equation

$$\dot{\mathbf{Y}}(t) = \frac{\partial \mathbf{f}}{\partial \mathbf{X}} \circ \mathbf{Y} \tag{2}$$

with a solution $Y(t) = (\xi, \dot{\xi})$ at time t, the maximum LE is given by

$$\gamma_1 = \lim_{t \to \infty} \frac{1}{t} \ln \frac{|\mathbf{Y}(t)|}{|\mathbf{Y}(0)|}.$$
 (3)

This is called as the variational method. The system is chaotic if $\gamma_1 > 0$, otherwise it is regular. This way is rigorous to get the tangent vector $\boldsymbol{Y}(t)$, but it is rather cumbersome to derive Eq. (2) in general. Because of that people usually use the deviation vector $\Delta \boldsymbol{X}(t)$ between a reference trajectory $\boldsymbol{X}(t)$ and a shadow one $\widetilde{\boldsymbol{X}}(t)$ as the approximate tangent vector. This is a less rigorous but still useful technique, so-called the two-particle method [18,24] or two-nearby-trajectories method, in which the LE in the expression of Eq. (3) is to be replaced by

$$\gamma_2 = \lim_{t \to \infty} \frac{1}{t} \ln \frac{|\Delta X(t)|}{|\Delta X(0)|}.$$
 (4)

Here an initial separation $|\Delta X(0)|$ relative to |X(0)| not larger than 10^{-8} is viewed as the best choice to guarantee ΔX as a good approximation to Y and to avoid the overestimation of LE [24].

As was mentioned in Ref. [23], it is preferable to compute the LE in the configuration space instead of in the phase space as LEs in the two spaces are both effective in detecting the long-term dynamical behavior of orbits.

In this case, Eqs. (3) and (4) are, respectively, modified as follows

$$\lambda_1 = \lim_{t \to \infty} \frac{1}{t} \ln \frac{|\boldsymbol{\xi}(t)|}{|\boldsymbol{\xi}(0)|},\tag{5}$$

$$\lambda_2 = \lim_{t \to \infty} \frac{1}{t} \ln \frac{|\Delta x(t)|}{|\Delta x(0)|}.$$
 (6)

For the two particles method, it is necessary to scale the distance $|\Delta \boldsymbol{x}(t)|$ down from time to time. In this way, the shadow trajectory returns to the neighborhood of the reference one along the deviation vector $\Delta \boldsymbol{x}(t)$. The magnitude of $\Delta \boldsymbol{x}(t)$ shrinks to the initial distance $|\Delta \boldsymbol{x}(0)|$ after each renormalization, and the velocity deviation vector $\Delta \dot{\boldsymbol{x}}(t)$ should be multiplied by the same factor $|\Delta \boldsymbol{x}(0)|/|\Delta \boldsymbol{x}(t)|$. Meanwhile, it is also vital to avoid saturation of orbits in a bounded chaotic region.

The time t and coordinate x in Eq. (5) or Eq. (6) are not necessary physical and meaningful in general relativity. Thus, it is desirable to define covariant LE.

B. LEs in general relativity

For a given 4-dimensional spacetime with the metric $ds^2 = g_{\mu\nu} dx^{\mu} dx^{\nu}$, first we consider a rigorous definition of LEs by using a geodesic deviation vector.

1. Geodesic deviation vector technique

In the spacetime, a free particle moves along the geodesic equation $DU^{\mu}/D\tau = 0$, that is to say,

$$\ddot{x}^{\mu} = -\Gamma^{\mu}_{\alpha\beta}\dot{x}^{\alpha}\dot{x}^{\beta},\tag{7}$$

and its geodesic deviation equation is

$$\frac{D^2 \xi^{\mu}}{D \tau^2} = -R^{\mu}_{\alpha \lambda \beta} \xi^{\lambda} U^{\alpha} U^{\beta}, \tag{8}$$

where $\Gamma^{\mu}_{\alpha\beta}$, $U^{\alpha}(=\dot{x}^{\alpha})$ and $R^{\mu}_{\alpha\lambda\beta}$ stand for the Christoffel symbol, 4-velocity, and the Riemannian curvature tensor, respectively.

From Eq. (5), it is easy to get an invariant definition of LE if the Riemannian norm and proper time τ substitute the Euclidian norm and coordinate time t, respectively. In the geodesic deviation vector method (M1) the LE is given in the form

$$\Lambda_1 = \lim_{\tau \to \infty} \frac{1}{\tau} \ln \frac{\|\boldsymbol{\xi}(\tau)\|}{\|\boldsymbol{\xi}(0)\|}, \tag{9}$$

$$\|\boldsymbol{\xi}(\tau)\| = \sqrt{|\boldsymbol{\xi} \bullet \boldsymbol{\xi}|} = \sqrt{|g_{\mu\nu}\xi^{\mu}\xi^{\nu}|}.$$
 (10)

As an illustration, a similar method given in the phase space can be seen in Ref. [28]. It is easily shown that the Riemannian inner product $\boldsymbol{\xi} \bullet \boldsymbol{\xi}$ is positive definite because $\boldsymbol{\xi}(\tau)$ is always space-like when $\boldsymbol{\xi}(0)$ is space-like for a given geodesic flow.

On the other hand, there is not a variational equation like the form (8) for a non-geodesic flow. However, it is still possible to derive the variational equation similar to Eq. (2) for the non-geodesic flow. For example, Nieto et al. [29] gave a relativistic top deviation equation as a generalization of the geodesic deviation equation for a pair of nearby point particles. Thus M1 remains valid if the spacetime background is $ds^2 = g_{\mu\nu} dx^{\mu} dx^{\nu}$.

2. Two-nearby-orbits method with projection operations

The derivation of the geodesic deviation equation is usually a rather hard task. In particular, it is much arduous to obtain the variational equation of a non-geodesic flow. Thus, it is a good idea to refine the classical definition by Eq. (6) using an invariant version of it, introduced by Wu and Huang [23] (method M2).

Let two particles, an "observer" and his "neighbor", move on two nearby trajectories in a curved spacetime. At a coordinate time t the observer is at the point O with coordinate x^{α} and 4-velocity U^{α} , and his neighbor reaches the point \widetilde{O} with coordinate \widetilde{x}^{α} . The deviation vector

$$\Delta x(t) = \Delta x^{\alpha}(t) = \widetilde{x}^{\alpha}(t) - x^{\alpha}(t) \tag{11}$$

from O to \widetilde{O} is projected to the observer's local space and the resulting projected vector is $\Delta x_{\perp}^{\alpha} = h_{\beta}^{\alpha} \Delta x^{\beta}$, where $h^{\alpha\beta} = g^{\alpha\beta} + U^{\alpha}U^{\beta}$ is the space projection operator of the observer. The space distance of the neighbor measured by the observer at time t is

$$\Delta L = \sqrt{g_{\alpha\beta} \Delta x_{\perp}^{\alpha} \Delta x_{\perp}^{\beta}} = \sqrt{h_{\alpha\beta} \Delta x^{\alpha} \Delta x^{\beta}}.$$
 (12)

Hence we define an invariant LE (M2) as

$$\Lambda_2 = \lim_{\tau \to \infty} \frac{1}{\tau} \ln \frac{\Delta L(\tau)}{\Delta L(0)},\tag{13}$$

where the proper time τ corresponds to the coordinate time t according to the metric. Obviously, this LE is independent of coordinate transformations [23].

There are detailed implementations of M2 in [23]. Here we emphasize that the coordinate time is adopted as the common independent variable for both particles in their equations of motion and one has to construct an equation for $d\tau/dt$, which is to be integrated together with the motion equations to get the proper time of the observer. The reason for doing this is that the two particles have their own and different proper times but the numerical integration demands a common time variable.

3. Two-nearby-orbits method without projection operations

On the other hand, if we integrate directly the equations of motion, Eq. (7), for two slightly distinct initial

conditions with the proper time as an integration variable, we attain the deviation vector

$$\Delta x(\tau) = \Delta x^{\alpha}(\tau) = \widetilde{x}^{\alpha}(\tau) - x^{\alpha}(\tau) \tag{14}$$

at the proper time τ of the observer. Here a difference between Eq. (11) and Eq. (14) should not be neglected. The difference is that Δx is a function of t in Eq. (11), while being a function of τ in Eq. (14). In this case, we can give another two-nearby-orbits method (M3) as follows

$$\Lambda_3 = \lim_{\tau \to \infty} \frac{1}{\tau} \ln \frac{\|\Delta x(\tau)\|}{\|\Delta x(0)\|},\tag{15}$$

$$\|\Delta \boldsymbol{x}(\tau)\| = \sqrt{|\Delta \boldsymbol{x} \bullet \Delta \boldsymbol{x}|} = \sqrt{|g_{\mu\nu}\Delta x^{\mu}\Delta x^{\nu}|}.$$
 (16)

Next let us study the three methods from the theoretical and numerical points of view.

C. The reason why M1 and M3 do not need projection operations

Obviously, projection operations do not appear in M1. Similarly, Hartl noted this fact in his PhD thesis [8]. Now, we give a discussion on this issue. It has been shown without any difficulty that $\boldsymbol{\xi}(\tau)$ is always perpendicular to the 4-velocity $\boldsymbol{U}(\tau)$ at any (proper) time provided that $\boldsymbol{\xi}(0)$ and $D\boldsymbol{\xi}(0)/D\tau$ are both perpendicular to $\boldsymbol{U}(0)$, respectively [28]. What would happen if $\boldsymbol{\xi}(0) \bullet \boldsymbol{U}(0) \neq 0$ or $D\boldsymbol{\xi}(0)/D\tau \bullet \boldsymbol{U}(0) \neq 0$? Let us explore the question in the following.

Noting that $\boldsymbol{\xi} \bullet \boldsymbol{U}$ and $D\boldsymbol{\xi}/D\tau \bullet \boldsymbol{U}$ are scalars, we can operate a covariant derivative in place of an ordinary one. The demonstration is

$$\frac{d}{d\tau}(\boldsymbol{\xi} \bullet \boldsymbol{U}) = \frac{D}{D\tau}(\boldsymbol{\xi} \bullet \boldsymbol{U}) = \frac{D\boldsymbol{\xi}}{D\tau} \bullet \boldsymbol{U} + \boldsymbol{\xi} \bullet \frac{D\boldsymbol{U}}{D\tau}$$

$$= \frac{D\boldsymbol{\xi}}{D\tau} \bullet \boldsymbol{U}, \qquad (17)$$

$$\frac{d}{d\tau} \left(\frac{D\boldsymbol{\xi}}{D\tau} \bullet \boldsymbol{U} \right) = \boldsymbol{U} \bullet \frac{D^2 \boldsymbol{\xi}}{D\tau^2} = -R^{\mu}_{\alpha\lambda\beta} \boldsymbol{\xi}^{\lambda} U^{\alpha} U^{\beta} U_{\mu}
= -R_{\nu\alpha\lambda\beta} U^{\alpha} U^{\beta} U^{\nu} \boldsymbol{\xi}^{\lambda}.$$
(18)

The result of Eq. (18) is obviously identical to zero because it is both symmetric and antisymmetric with respect to the indicies ν and α . Consequently, it can be inferred that the right side of Eq. (17) is a constant C, which turns out to be the value of $D\xi(0)/D\tau \bullet U(0)$. As was stated above, $\xi \bullet U \equiv 0$ if C = 0. Otherwise we have

$$\boldsymbol{\xi} \bullet \boldsymbol{U} = C\tau + \widetilde{C},\tag{19}$$

where \widetilde{C} is another constant corresponding to the initial value of $\boldsymbol{\xi} \bullet \boldsymbol{U}$. Now let us investigate the physical significance of $|\boldsymbol{\xi} \bullet \boldsymbol{U}|$. Obviously it is no other than the length of the projected vector $-(\boldsymbol{\xi} \bullet \boldsymbol{U})\boldsymbol{U}$, the time component of $\boldsymbol{\xi}$ as measured by the particle. Equation (19)

shows that the evolution of $\boldsymbol{\xi}$ along the time direction of the particle is linear, therefore, the LE in this direction vanishes. In fact, the result is very natural due to the integral $U_{\alpha}U^{\alpha}=-1$. Therefore, $\boldsymbol{\xi}$ and its projected vector along the space direction of the particle change at the same rate with its proper time. This implies that it is not necessary to project $\boldsymbol{\xi}$ into the space direction in the calculation of the principal LE. In other words, it is reasonable to use M1 as the standard definition of the maximum LE in general relativity.

As to M3, the deviation vector $\Delta x(\tau)$ in Eq. (14) is viewed as an approximation to the geodesic deviation vector $\boldsymbol{\xi}$ by Eq. (8). Therefore, M3 is nearly the same as M1 as long as the length of $\boldsymbol{\xi}$ is kept small enough.

For M2, we use the projected vector $\Delta x_{\perp}^{\alpha}$ rather than $\Delta x(t)$ used in Eq. (11) as an approximation of $\boldsymbol{\xi}$. $\Delta x(t)$ in Eq. (11) is the deviation vector between the observer and his neighbor at the same coordinate time t, therefore it is necessary for $\Delta x(t)$ to be projected along the space direction of the observer. On the other hand, $\Delta x(t)$ and \boldsymbol{U} do not necessarily obey the relation (19) when t and τ have large difference. Then in the observer's in-track direction he may find LE in the form

$$\lambda_{in\text{-}track} = \lim_{\tau \to \infty} \frac{1}{\tau} \ln \frac{\Delta T(\tau)}{\Delta T(0)},\tag{20}$$

where $\Delta T(\tau) = |\Delta x(t) \bullet U|$ is the projection of $\Delta x(t)$ along the observer's time direction. This shows that one may find chaos in the observer's time direction if $\lambda_{in\text{-}track}$ is used as an index for chaos. This displays it necessary to employ the projection norm for M2.

We shall further check the validity of the three methods through practical calculations.

D. Numerical comparisons

Let us take a Weyl spacetime as a physical model (see the appendix), and choose parameters as E = 0.9679, L = 3.8 and $\mathcal{O} = 7.012 \times 10^{-7}$. The initial conditions of two variables r and \dot{r} are arbitrarily given within their respective admissible intervals except $\theta = \pi/2$, while θ is derived from Eq. (A4). First order differential systems from the geodesic Eqs. (A5) and (A6) are computed with a 7-8 order Runge-Kutta-Fehlberg algorithm of variable time-step (RKF7(8)). A proper time output interval is 0.1, and the Poincaré surface of section is at the plane $\theta = \frac{\pi}{2}$ ($\dot{\theta} < 0$) in Fig. 1. The intersections of the orbits by this Poincaré surface of section describe the global dynamical feature of the system. There are two regions in the (r, \dot{r}) plane. One region with randomly distributed points is regarded as a chaotic one, while the other, a regular one consisting of many tori and islands.

Now we quantitatively describe the dynamics of particles moving in this spacetime by employing the LEs above. For M1, it is necessary to integrate the dynamical Eqs. (A2)-(A6) numerically with their geodesic deviation equations together so that we obtain their solu-

tion (x^{α}, U^{α}) and their variational solution $(\xi, \dot{\xi})$ in the forms $\xi = (\delta t, \delta r, \delta \theta, \delta \phi)$ and $\dot{\xi} = (\delta \dot{t}, \delta \dot{r}, \delta \dot{\theta}, \delta \dot{\phi})$. Here the proper time τ is chosen as an integration variable. In addition, the renormalization technique is used if necessary. As far as M2 is concerned, we numerically trace the trajectories of the observer and his neighbor with the Eqs. (A7,A8) and the equations involving $d\tau/dt$ and $d\phi/dt$ with slightly different initial conditions. The initial conditions of the observer are the same as the above. As to his neighbor, an initial separation in the order of 10^{-8} is adopted only on the r direction, regarded as the best choice [24], and the other coordinates remain the same as the observer's except $\dot{\theta}$. This process also fits M3, but only Eqs. (A2)-(A6) are integrated numerically.

Choosing many ordered and chaotic orbits as numerical tests, we found that the three methods M1, M2 and M3 almost give the same final values of LEs for each orbit. There are only a few differences on the transition phases of the orbits. For example, we choose a chaotic orbit with the initial conditions: t=0, $\theta=\pi/2$, r=9, $\phi=0$, $\dot{r}=0.025$ and the initial value of $\dot{\theta}$ derived from Eq. (A4). Let the initial deviation vector satisfy $\delta t=\delta\theta=\delta\phi=\delta\dot{r}=\delta\dot{\theta}=0$ and $\delta r=1/\sqrt{g_{11}}$. As is shown in Fig. 2, M3 is always close to M1 (that is, M3 is just a good approximation of M1), but M2 is not consistent with M1 until the stabilizing value of LEs is obtained. Consequently, the three methods, M1, M2 and M3, have almost the same final value of LEs.

For conceptual clarity it is of the physical significance to define LE as a coordinate gauge invariant in general relativity. For this purpose, there are the above three methods (M1, M2 and M3) to compute LEs. M1 is more rigorous than M2 or M3. On the contrary, M2 is more convenient to use than M1, and M3 is the simplest. In addition, M2 and M3 are easier to treat a non-geodesic flow as well as a geodesic flow than M1. Numerical experiments display that M1, M2 and M3 not only can obtain the same final values but also have small differences in the transition phase for the calculation of LEs. Considering the application of convenience, we conclude that M3 should be worth recommending to calculate LEs in a relativistic gravitational system. It is useful to accept some advice on a suitable choice of the initial separation and a renormalization time step introduced by Tancredi et al [24].

Although the LEs have been widely used to distinguish between regular and chaotic orbits, their stabilizing values obtained often need rather expensive computational cost in exploring the global structure of phase space. In view of this, we shall modify M3 and adopt its corresponding FLI in the following section.

III. DESCRIPTION OF THE STRUCTURE OF PHASE SPACE BY USING FLI WITH TWO-NEARBY-TRAJECTORIES

A. Fast Lyapunov indicators

The time interval that is necessary to reach a given value of either the length of a tangential vector or the angle between two tangential vectors can be taken as an indicator of stochasticity of a Newtonian dynamical system. Following this idea, Froeschlé $et\ al.$ [21] defined three different FLIs in 1997. However two of these methods need to solve the variational Eq. (2) for n times when a set of n independent initial tangential vectors with the same initial conditions are chosen. In this circumstance, Froeschlé and Lega [22] improved the FLI as follows

$$\psi(t) = \ln |\mathbf{Y}(t)|,\tag{21}$$

where Y is a tangential vector from Eq. (2) and |Y(0)| = 1. Given a threshold, the indicator ψ reaches a value fast for a chaotic orbit, but it would take a rather long time for an ordered one. Conversely, in the same time interval the indicator tends to different values for ordered and chaotic orbits. Namely it grows exponentially with time for the chaotic orbit, but only algebraically with time in the regular case. This allows one to distinguish between the two cases. There is a close relation between the FLI (ψ) and the LE [Eq. (3)]: the FLI divided by time t tends to the LE when the time is sufficiently large. Besides, overflow of the lengths of tangential vectors in the case of a chaotic orbit can be avoided because the integration time is not long enough. This is the reason why the indicator is classified as a "fast" method.

Following the idea above and the coordinate gauge invariance, in a curved spacetime we could easily define an invariant FLI corresponding to the LE in Eqs. (9) and (15). As far as the geodesic deviation vector method M1 is concerned, its corresponding FLI is

$$\Psi(\tau) = \log_{10} \|\boldsymbol{\xi}(\tau)\|. \tag{22}$$

In the light of Eq. (15), we define the FLI with twonearby-trajectories as follows:

$$FLI(\tau) = \log_{10} \frac{\|\Delta \boldsymbol{x}(\tau)\|}{\|\Delta \boldsymbol{x}(0)\|}.$$
 (23)

Next we shall mainly describe the numerical implementation of the FLI and test its validity in the description of dynamics.

B. The algorithm in detail

The original FLI proposed by Froeschlé and Lega [22] requires to compute the expansion rate of a tangential vector in the variational equations and it does not need any renormalization. We now reform it in general relativity into a coordinate invariant and compute it with the

two particles approach. The $\mathrm{FLI}(\tau)$ in Eq. (23) can not be computed without renormalization because the distance between the two particles, $\|\Delta x\|$, would expand so fast in the case of chaos as to it could reach the chaotic boundary to cause saturation.

In our numerical model (see the appendix) we choose $\|\Delta x(0)\| = 10^{-9}$. We found saturation when $\|\Delta x\| = 1$, therefore we choose $\|\Delta x\| = 0.1$ as the critical value to implement the renormalization. In this way, the number of renormalization for computing FLI is less than that for LE. This brings an advantage to guarantee the speed of computation. Let k ($k = 0, 1, 2, \cdots$) be the sequential number of renormalization, then we calculate the FLI with the following expression

$$FLI_{k}(\tau) = -k \cdot (1 + \log_{10} \|\Delta \boldsymbol{x}(0)\|) + \log_{10} \frac{\|\Delta \boldsymbol{x}(\tau)\|}{\|\Delta \boldsymbol{x}(0)\|},$$
(24)

where $\|\Delta x(0)\| \leq \|\Delta x(\tau)\| \leq 0.1$. This technique depends on the choice of the initial deviation $\|\Delta x(0)\|$, and fails when $\|\Delta x(0)\|$ is too small or too large. After many numerical experiments, we found that $\|\Delta x(0)\| \approx 10^{-7} - 10^{-9}$ works well.

C. Numerical tests of FLI

We take the initial separation $\Delta r = 10^{-9}$ and choose the regular orbit M and the chaotic one N in Fig. 1 to test the sensitivity of the FLI as well as Ψ . As is expected, Fig. 3 (colored light gray) displays that there is a drastic difference of the FLIs between the regular orbit and the chaotic one. For the ordered orbit, within a time span of 10⁵ (equivalent to about 100 periods), the FLI is smaller than one. However, we clearly see a sharp increase up to 24 of the FLI for the chaotic orbit. In other words, the length of the geodesic deviation vector ξ reaches the value 10^{24} . As mentioned above, the vector increases in an algebraic law for quasi-periodic orbits, but it does with an exponential law for chaotic orbits. Indeed the chaos of the orbit in the right of Fig. 3 becomes explicit after a time span 4.5×10^4 . The FLI is a cheaper way to distinguish between chaotic and regular orbits and explore the global qualitative structure of the phase space of a system.

Another point to emphasize is that the values of FLI are very analogous to that of Ψ (see black dots in Fig. 3). The values of FLIs by use of the two methods for the regular case have no difference in a long time since renormalization is not used. For the chaotic case we use only two times of renormalization. In general, it is enough to take small numbers of renormalization to calculate the FLIs in the orbits with weak chaos. This shows sufficiently the validity of our FLI with two-nearby-trajectories.

To examine the validity of the FLI further, in the left of Fig. 4 we have plotted the running maximum of FLIs as a function of the initial action r for a set of 1657 orbits that are regularly spaced in the interval [5.77, 22.34]

on the axis $\dot{r} = 0$ in Fig. 1. Here each orbit is integrated until the time reaches 10^5 . As is described in Fig. 1, the regular region is located in one interval between R(8.45,0) and S(21,0) and another interval between R(22.15,0) and the point (22.34,0) on the straight line $\dot{r} = 0$. Obviously we find that all the FLIs in the two intervals are not larger than 6, but all the FLIs in the two intervals [5.77, 8.45] and [21, 22.14] on the straight line $\dot{r}=0$ are larger. This method displays the set of chaos. although it does not give the final values of LEs. In a similar way, the orbits along the straight line r=9 in Fig. 1 are also separated into regular and chaotic intervals (see the right of Fig. 4). These facts show sufficiently that the FLI is rather successful to distinguish between the regularity and the chaoticity. However, we must point out a problem that the FLI does not provide more details about ordered orbits containing quasi-periodic and resonant periodic orbits. Here we give an illustration in detail.

On the basis of Froeschlé and Lega [22], the orbit with the starting point M(15,0) in Fig. 1 corresponds to a resonance since the maximum value of FLI arrives at about 0.68 as the smallest value in the left of Fig. 4. However this is not a resonant periodic orbit but just a quasiperiodic one. Actually the point I(14.00349, -0.0312004)of the Poincaré map in the middle of tori is a fixed point. We found that all tori between the torus Mand the point I have almost the same values of FLI. We cannot say that FLI is appropriate to identify the resonance categories. As another example, let us focus a trajectory made up of three little loops or islands surrounding three invariant points $J_1(9.0975, -0.0281)$, $J_2(17.13931, 0.04913)$ and $J_3(13.94246, -0.08936)$ in Fig. 1, respectively. An important point to mention lies in the manner where the three little loops appear on the plot as the same trajectory. Rather than tracing one loop at a time, successive points occur at each of the three loops in turn. In this sense, the three fixed points are inhabited in the same periodic orbit. The maximum value of FLI we computed for this periodic (resonant) orbit after a time span of 10^5 turns out to be 1.75 or so. Obviously the value of FLI for the resonant orbit is larger than that for the quasi-periodic orbit M. In a word, it is difficult to apply values of FLI to distinguish various regular orbits.

D. Classification of orbits by FLI

Many experiments above have shown that our FLI is a very sensitive tool for detecting the regular and chaotic orbits. Next we shall follow this FLI to scan the global structure of phase space in the spacetime (A1). This operation is realized by calculating thousands of orbits in practice. First we fix $\theta=0$, meanwhile we let r run from $r_{min}=5.77$ to $r_{max}=22.34$ with a span of $\Delta r=0.1$. Then, for each given r we can solve for \dot{r} and find two roots \dot{r}_- and \dot{r}_+ (\dot{r}_- < 0 and \dot{r}_+ > 0) from the Eq. (A4). Finally we take \dot{r} from \dot{r}_- to \dot{r}_+ with a sampling

interval $\Delta \dot{r} = 0.01$. Once r and \dot{r} are given initially, $\dot{\theta}$ should be derived by the relation (A4). As is shown in Fig. 5, we have plotted all the starting points on the $r - \dot{r}$ plane ($\theta = \pi/2$). According to different values of FLI, we classify orbits by some dynamical features. This is called as the description of the global structure of phase-space for the system.

By a comparison of Fig. 5 with Fig. 1, it is clear that the global dynamical features they depicted are nearly compatible. As an emphasis, an advantage to use FLI becomes more apparent because it applies to systems with an arbitrary number of dimensions. Without doubt, it is handy to study the global dynamics of the spinning compact binaries [10] by virtue of FLI. It is also easier to probe the variation of the dynamical characteristics as certain parameters of the system vary.

IV. SUMMARY

We compared three methods for computing LEs with coordinate gauge invariance in general relativity. The three methods are the geodesic deviation vector technique (M1), the two-nearby-orbits method with projection operations and with coordinate time as the integration independent variable (M2), and the two-nearby-orbits method without projection operations and with proper time as the integration independent variable (M3). The contributions of this work are as follows.

It is unnecessary to adopt the projection operators for M1 and M3 from the theoretical point of view. Method M3 is the simplest method for calculating LEs in a relativistic gravitational system in most cases. As another contribution, we extended FLI to a coordinate invariant form in relativistic dynamics, and proposed its algorithm with the two-nearby-trajectories method. This indicator can rapidly, reliably and accurately distinguish between ordered and chaotic motions in relativistic astrophysics. Only when the initial separation and renormalization within a reasonable amount of time span are chosen appropriately, is this FLI nearly consistent with the FLI computed with the geodesic deviation vector technique. However, the former is rather simpler to use than the latter. As a characteristic, our FLI grows exponentially with time in the chaotic case, while it grows algebraically in the case of quasi-periodic trajectories. Evaluating the different behaviors of this FLI, we successfully explored the global dynamics of phase space where regions of chaos and order are clearly identified.

It should be pointed out that a main advantage of the LE and FLI with two-nearby-trajectories is their easier application in treating complicated relativistic gravitational systems with high degrees of freedom, such as multi-body problems and spinning compact binaries. In the future work, we shall employ an appropriate numerical integrator and the FLI to find out what happens in the spinning compact binaries. The global dynamics of the binary systems will be explored, and the transition

from regular motion to chaos will be considered as some dynamical parameters of the system vary.

Appendix A: Core-shell system and the equations of motion

A Weyl spacetime including a non-rotating black hole surrounded by an axially symmetric shell in Schwarzchild coordinates (t, r, θ, ϕ) is expressed as [6]

$$ds^{2} = -(1 - \frac{2}{r})e^{P}dt^{2} + e^{Q-P}[(1 - \frac{2}{r})^{-1}dr^{2} + r^{2}d\theta^{2}] + e^{-P}r^{2}\sin^{2}\theta d\phi^{2},$$
(A1)

where P and Q are the functions of r and θ only. This system obviously has two integrals, the energy E and the angular momentum L in the forms

$$\dot{t} = Er(r-2)^{-1}e^{-P},$$
 (A2)

$$\dot{\phi} = Le^P r^{-2} \sin^{-2} \theta. \tag{A3}$$

In addition, the 4-velocity of a particle always satisfies the constraint

$$U_{\alpha}U^{\alpha} = -(1 - \frac{2}{r})e^{P}\dot{t}^{2} + e^{Q-P}[(1 - \frac{2}{r})^{-1}\dot{r}^{2} + r^{2}\dot{\theta}^{2}] + e^{-P}r^{2}\sin^{2}\theta\dot{\phi}^{2} = -1.$$
 (A4)

If a fourth integral holds, the system is integrable. Otherwise, it is possible to yield chaos. However, it is not easy to deduce whether the fourth integral exists or not. To study the dynamics of a geodesic particle in this system further, we need the following geodesic equations

$$\ddot{r} = \left[\frac{1}{r(r-2)} + \frac{1}{2} \left(\frac{\partial P}{\partial r} - \frac{\partial Q}{\partial r}\right)\right] \dot{r}^2 - \left(\frac{\partial Q}{\partial \theta} - \frac{\partial P}{\partial \theta}\right) \dot{r} \dot{\theta}$$

$$+ (r-2)\left[1 + \frac{r}{2} \left(\frac{\partial Q}{\partial r} - \frac{\partial P}{\partial r}\right)\right] \dot{\theta}^2 + f_1(r,\theta), \quad (A5)$$

$$\ddot{\theta} = \frac{1}{2r(r-2)} \left(\frac{\partial Q}{\partial \theta} - \frac{\partial P}{\partial \theta}\right) \dot{r}^2 - \left(\frac{2}{r} + \frac{\partial Q}{\partial r} - \frac{\partial P}{\partial r}\right) \dot{r} \dot{\theta}$$

$$- \frac{1}{2} \left(\frac{\partial Q}{\partial \theta} - \frac{\partial P}{\partial \theta}\right) \dot{\theta}^2 + f_2(r,\theta), \quad (A6)$$

where

$$f_1(r,\theta) = -\frac{1}{2}(1 - \frac{2}{r})e^{2P - Q}\left[\frac{2}{r^2} + (1 - \frac{2}{r})\frac{\partial P}{\partial r}\right]\dot{t}^2$$
$$+\frac{1}{2}(r - 2)e^{-Q}\sin^2\theta \cdot (2 - r\frac{\partial P}{\partial r})\dot{\phi}^2,$$
$$f_2(r,\theta) = -\frac{1}{2r^2}(1 - \frac{2}{r})e^{2P - Q}\frac{\partial P}{\partial \theta} \cdot \dot{t}^2 + \frac{1}{2}e^{-Q}$$
$$\cdot \sin\theta \cdot (-\frac{\partial P}{\partial \theta}\sin\theta + 2\cos\theta)\dot{\phi}^2.$$

We now have to do the tedious derivation of the geodesic deviation equations. The result is too complex and will not be written here. If we take a' as the derivative of a with respect to coordinate time t, the above geodesic equations are readjusted as follows

$$r'' = \frac{\ddot{r}}{\dot{t}^2} - \frac{r'}{\dot{t}} \frac{d\dot{t}}{dt}, \tag{A7}$$

$$\theta'' = \frac{\ddot{\theta}}{\dot{t}^2} - \frac{\theta'}{\dot{t}} \frac{d\dot{t}}{dt}, \tag{A8}$$

$$\frac{d\dot{t}}{dt} = -Er(r-2)^{-1} e^{-P} (\frac{\partial P}{\partial r} r' + \frac{\partial P}{\partial \theta} \theta')$$

Here \dot{r} and $\dot{\theta}$ in Eqs. (A5) and (A6) should be changed into $\dot{r} = r'\dot{t}$ and $\dot{\theta} = \theta'\dot{t}$ respectively, while \dot{t} and $\dot{\phi}$ remain the original expressions of Eqs. (A2) and (A3) because they do not contain coordinate time t explicitly.

 $-2E(r-2)^{-2}e^{-P}r'$.

Given two metric functions P and Q, the spacetime (A1) is determined at once. Now we reexamine the full relativistic core-shell configuration involving a Schwarzschild black hole plus a pure octopole shell studied by Vieira & Letelier [6]. In this model, we have

$$P(u,v) = \frac{1}{5}\mathcal{O}uv(5u^2 - 3)(5v^2 - 3),$$

$$Q(u,v) = -\frac{2}{5}\mathcal{O}v[5(3u^2 - 1)(1 - v^2) - 4]$$

$$+\frac{3}{100}\mathcal{O}^2[-25u^6(1 - v^2)$$

$$\cdot (5v^2 + 2v - 1)(5v^2 - 2v - 1)$$

$$+15u^4(1 - v^2)(65v^4 - 40v^2 + 3)$$

$$-3u^2(1 - v^2)(25v^2 - 3)(5v^2 - 3)$$

$$-v^2(25v^4 - 45v^2 + 27)],$$

where u = r - 1, $v = \cos \theta$ and \mathcal{O} is an octopolar parameter.

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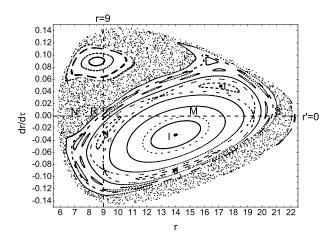


FIG. 1: Poincaré surface of section at the plane $\theta=\frac{\pi}{2}$ ($\dot{\theta}<0$) for the full relativistic core-shell configuration consisting of a Schwarzschild black hole and a purely octopolar shell with parameters $E=0.9679,\,L=3.8$ and $\mathcal{O}=7.012\times10^{-7}.$

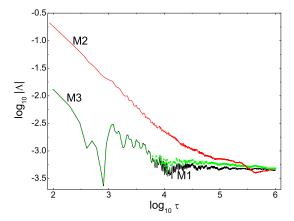
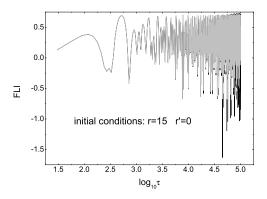


FIG. 2: LEs for a chaotic orbit with the initial conditions r=9 and $\dot{r}=0.025$. The three methods, M1, M2 and M3, give almost the same final value of LEs.



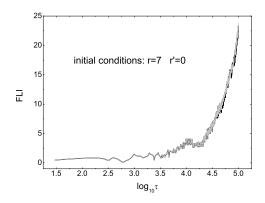
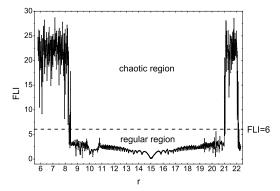


FIG. 3: The relation of FLIs with proper time by use of the index (23) for the regular orbit M in the left (colored light gray). The black dots correspond to the result obtained from Eq. (22). The right is the same as the left but for the chaotic orbit N in Fig. 1.



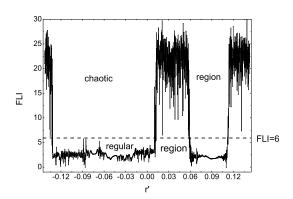


FIG. 4: Left: the evolution of the FLIs with the initial values of r ranging from the interval [5.77, 22.34] when the indicator (23) is used and $\dot{r}=0$ is fixed at the initial time. Each orbit is integrated numerically till τ reaches 10^5 . Two types of orbits consisting of regular and chaotic orbits are obtained according to distinct values of FLIs. Right: the same as the left, but r=9 is fixed first, and \dot{r} takes values from the interval [-0.1428, 0.1428].

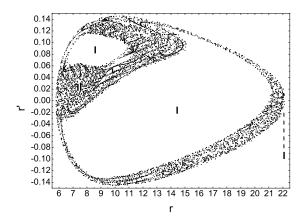


FIG. 5: Regions of different values of the FLIs on the plane $\theta=\frac{\pi}{2}$. Initial conditions are colored white if their FLIs≤6, and black if FLIs>6, which are corresponded to the regular region I, and chaotic region II, respectively. Here, all intersection points of an orbit with the plane should be regarded to have the same final value of the FLI.